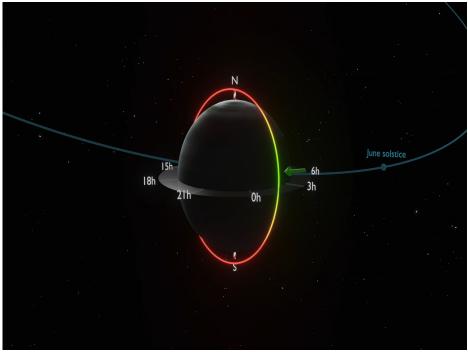


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# Daedalus – a candidate for the 10th ESA Earth Explorer mission





For more information: Sarris et al., 2020 https://doi.org/10.5194/gi-9-153-2020

Table 2. List of Daedalus instruments, measurements, estimated dynamic ranges, accuracies and sensitivities.			https://doi.org/10.5194/gi-9-153-2020
Instrument	Measurement	Dynamic range	Accuracy, sensitivity
Ion Drift Meter (IDM) and Retarding Potential Analyzer	Ion drifts Ion density	$\pm 4\mathrm{kms^{-1}}$ (along-track and cross-track)	100 m s <sup>−1</sup> (along-track and cross-track)

Ion Drift Meter (IDM) and Retarding Potential Analyzer (RPA) or Thermal Ion Imager (TII)	Ion drifts Ion density Ion temperature	$\pm 4\mathrm{kms^{-1}}$ (along-track and cross-track)	$100\mathrm{ms^{-1}}$ (along-track and cross-track)
Ram Wind Sensor (RWS) and Cross-track Wind Sensor (CWS)	Ram neutral winds Cross-track neutral winds Differential pressure Neutral temperature	$\pm 1\mathrm{km}\mathrm{s}^{-1}$ (along-track and cross-track)	Accuracy $\pm 10 \mathrm{ms^{-1}}$ , sensitivity $\pm 3 \mathrm{ms^{-1}}$ (along-track) Accuracy $\pm 5 \mathrm{ms^{-1}}$ , sensitivity $\pm 2 \mathrm{ms^{-1}}$ (cross-track)

Ram Wind Sensor (RWS) and Cross-track Wind Sensor (CWS)	Ram neutral winds Cross-track neutral winds Differential pressure Neutral temperature	$\pm 1\mathrm{km}\mathrm{s}^{-1}$ (along-track and cross-track)	Accuracy $\pm 10 \mathrm{ms^{-1}}$ , sensitivity $\pm 3 \mathrm{ms^{-1}}$ (along-track) Accuracy $\pm 5 \mathrm{ms^{-1}}$ , sensitivity $\pm 2 \mathrm{ms^{-1}}$ (cross-track)
Accelerometer (ACC)	Neutral density Wind velocity Thrust of propulsion syst.	$10^{-7}g$ to $10^{-3}g$	Accuracy $\pm 10\%$ at 500 km, $\pm 2\%$ below 200 km Sensitivity $10^{-7}g$ , $\pm 3\%$ max systematic error due

	redutal temperature		
Accelerometer (ACC)	Neutral density Wind velocity Thrust of propulsion syst.	$10^{-7}g$ to $10^{-3}g$	Accuracy $\pm 10\%$ at $500$ km, $\pm 2\%$ below $200$ km Sensitivity $10^{-7}g$ , $\pm 3\%$ max systematic error due to uncertainty in drag coefficient

	Thrust of propulsion syst.		±3 % max systematic error due to uncertainty in drag coefficient
Energetic Particle Detector Suite (EPDS), including the High Energy Instrument	HEI: relativistic electrons, protons, heavy ions LEI: low-energy electrons, ions	HEI: $10^{1}$ – $10^{0}$ counts per second LEI: $10^{6}$ – $5 \times 10^{9}$ eV $(\text{cm}^{2} \text{ sr s eV})^{-1}$	HEI: accuracy $\leq$ 20 % LEI: accuracy $\leq$ 20 % for electron energy fluxes above $10^6$ eV (cm <sup>2</sup> sr s eV) <sup>-1</sup>
(HEI), Low Energy Instrument (LEI)	ENA: energetic neutral atoms	ENA: energies 5–200 keV, fluxes $10^2$ –2 × $10^6$ (cm <sup>2</sup> sr s) <sup>-1</sup>	ENA: energy resolution of at least 15 keV, flux to better than 20 % for fluxes above 2000 (cm $^2$ sr s) $^{-1}$

HEI: 
$$10^1-10^0$$
 counts per second LEI:  $10^6-5\times 10^9$  eV LEI: accuracy  $\leq 20\,\%$  for electron energy fluxes above  $10^6$  eV (cm<sup>2</sup> sr s eV)<sup>-1</sup> (cm<sup>2</sup> sr s eV)<sup>-1</sup> ENA: energies  $5-200\,\text{keV}$ , fluxes  $10^2-2\times 10^6$  (cm<sup>2</sup> sr s)<sup>-1</sup> ENA: energy resolution of at least  $15\,\text{keV}$ , flux to better than  $20\,\%$  for fluxes above  $2000\,\text{(cm}^2\,\text{sr s})^{-1}$ 

the High Energy Instrument (HEI), Low Energy Instrument (LEI) and Energetic Neutral Atom (ENA) instrument	protons, heavy ions LEI: low-energy electrons, ions ENA: energetic neutral atoms	LEI: $10^{6}-5 \times 10^{5} \text{ eV}$ $(\text{cm}^{2} \text{ sr s eV})^{-1}$ ENA: energies 5–200 keV, fluxes $10^{2}-2 \times 10^{6} \text{ (cm}^{2} \text{ sr s)}^{-1}$	LEI: accuracy $\leq 20\%$ for electron energy fluxes above $10^{\circ}$ eV (cm <sup>2</sup> sr s eV) <sup>-1</sup> ENA: energy resolution of at least 15 keV, flux to better than 20% for fluxes above 2000 (cm <sup>2</sup> sr s) <sup>-1</sup>
Ion Mass Spectrometer (IMS) and Neutral Mass Spectrometer (NMS)	Ion composition (IMS) Neutral composition (NMS) Relative density	Mass range: 1–50 amu Ions: H <sup>+</sup> , He <sup>+</sup> , N <sup>+</sup> , O <sup>+</sup> , NO <sup>+</sup> , $O_2^+$ , $CO_2^+$ Neutrals: H, He, N, O, N <sub>2</sub> , NO, CO <sub>2</sub> Density dynamic range: Ions: $\sim 10^2 - 10^7$ cm <sup>-3</sup> Neutrals: $\sim 10^4$ to $10^{13}$ cm <sup>-3</sup>	Mass resolution accuracy M / dM: $\sim$ 30 Mass resolution sensitivity: 1 amu Relative density resolution accuracy: 1 %–10 % (TBD) Relative density resolution: 1 %

Temperature range: 200–2000 K

### Dual-satellite option under investigation

Vertical profiles at satellite latitude and longitude

Thin red/blue lines: TIE-GCM 2.0

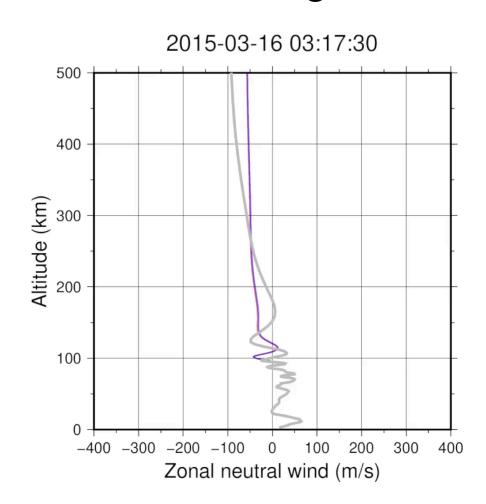
Grey line: WACCM-X 2.1

Red/blue dots: TIE-GCM sampled at satellite

altitude

Red: Satellite A (perigee 150 km) Blue: Satellite B (perigee 200 km)



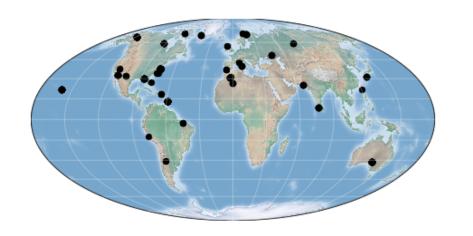


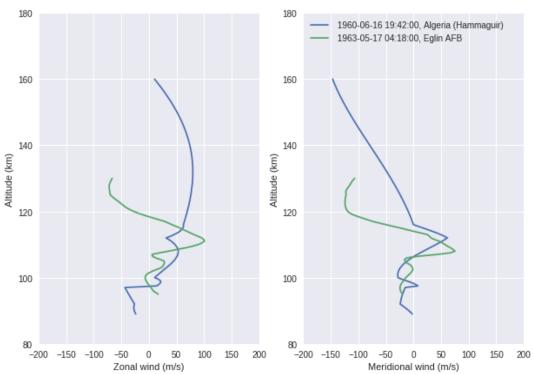
### Research questions

- 1. Identify the overlap and gaps in coverage of in-situ wind data at 100-250 km altitude
- 2. What are the highest horizontal winds and shears that can be found in this data? When and where can they be expected?
- 3. What are the advantages and limitations of the use of models (in this case HWM14) in the mission definition.

## Sounding rocket chemical tracer (CT) data

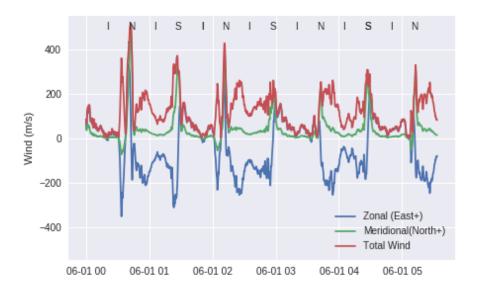
- Data courtesy of Miguel Larsen (Clemson Univ)
- 565 runs (1955-2009)
- 35 locations

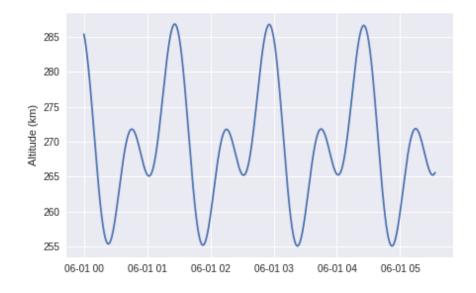




#### GOCE accelerometer data:

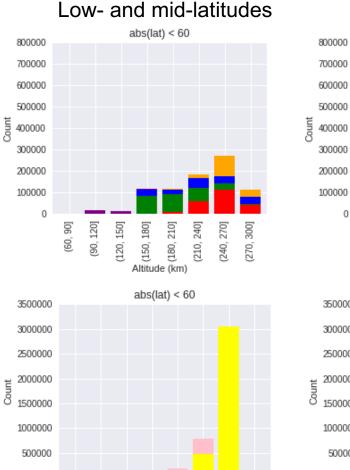
- Cross-track wind derived from accelerations
- Science phase
  - Nov 2009-Oct 2013
  - ~230-270 km
- Deorbit phase
  - 22 Oct 2013-11 Nov 2013
  - ~170-230 km





# Altitude coverage

Coverage data of HWM14 database, courtesy of John Emmert and Manbharat Dhadly, NRL



(150, 180]

(180, (210, (240,

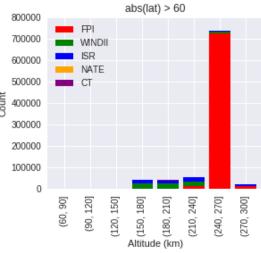
(120, 150]

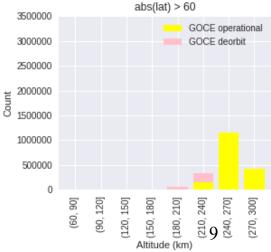
240]

270]

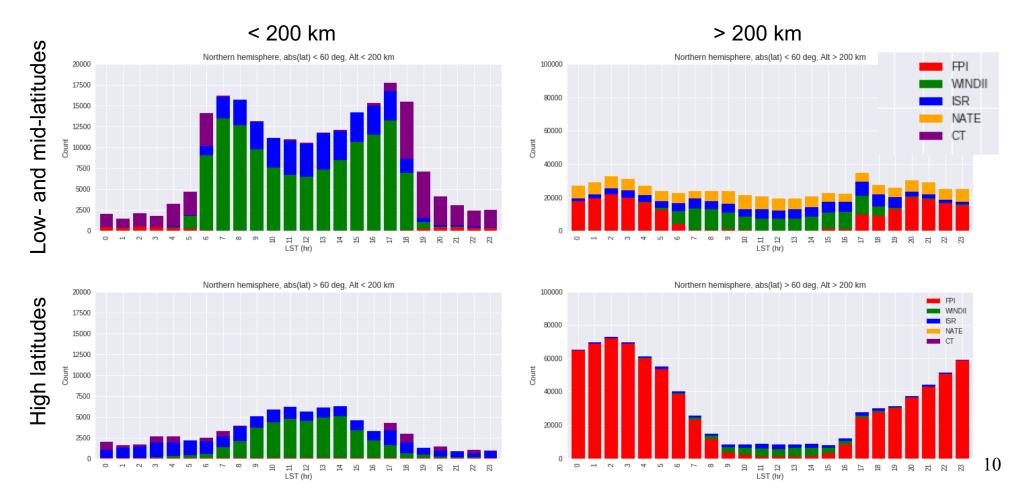
(60, 90] (90, 120]



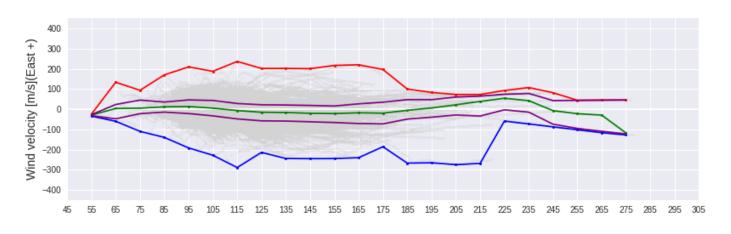


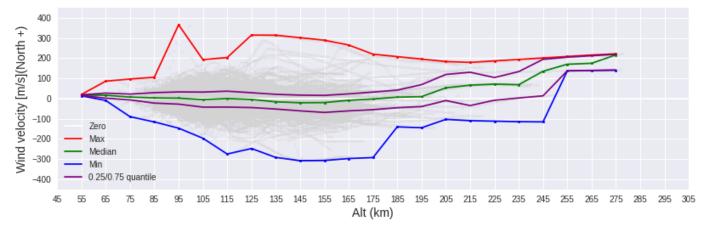


# LST coverage: Northern Hemisphere

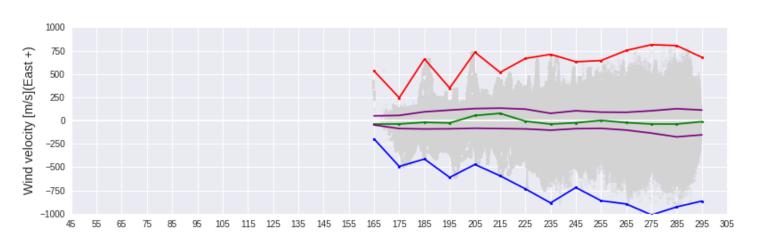


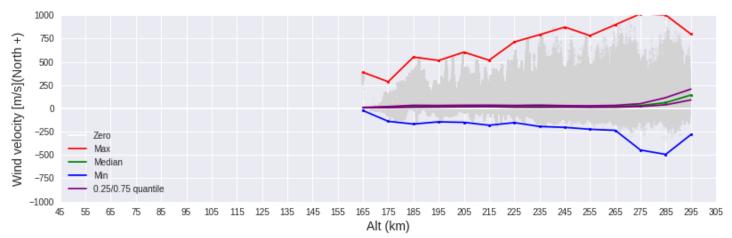
# Altitude dependency sounding rocket chemical tracer



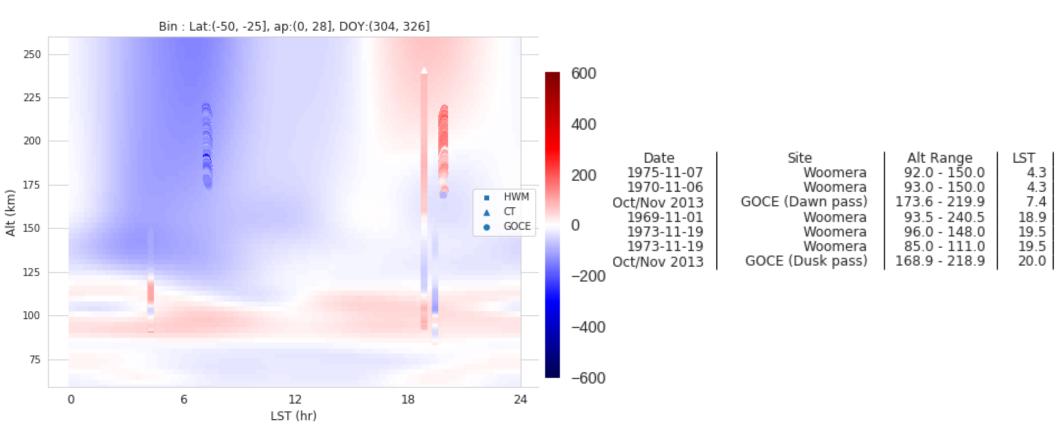


## Altitude dependency GOCE crosswind





## Both data types compared with HWM14

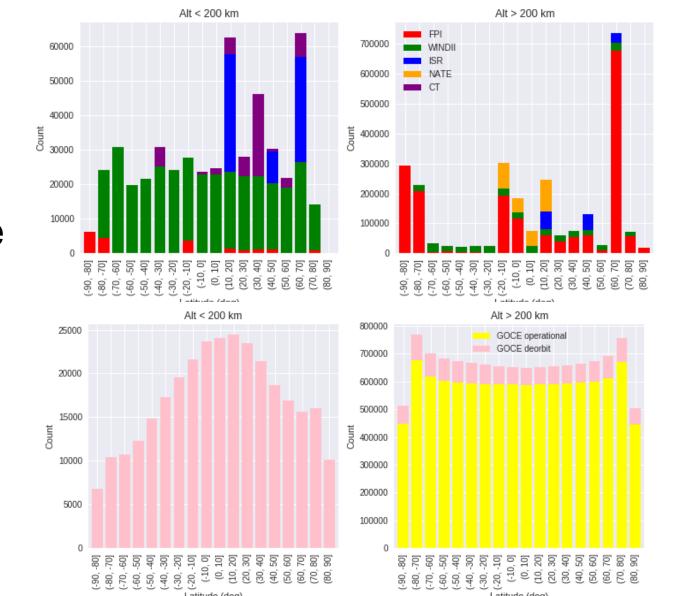


#### Conclusions

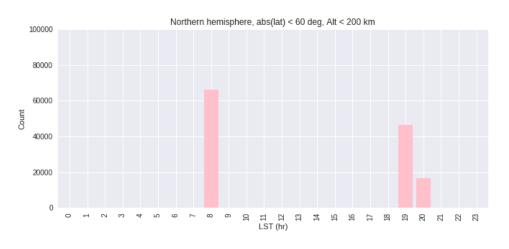
- The sounding rocket chemical tracer data provides unique low-altitude height-resolved measurements in the lower thermosphere, providing information on vertical structure, that is useful for planning the Daedalus mission
- The GOCE deorbit dataset contains much higher winds (up to ~800 m/s) at high latitudes, not found in the sounding rocket data (max ~400 m/s) or HWM14
- The CT dataset contains higher shears at low altitudes which are not found in HWM14
- Similarities in vertical structure found by sounding rockets, GOCE and HWM14
- If selected by ESA, Daedalus could bring unique in-situ wind data in the lower thermoshere, that will be height-resolved, and covers all lat/lon locations and a wide range of season/local solar time conditions

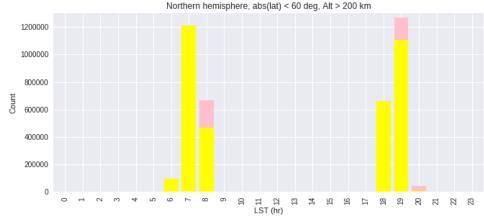
# Back-up slides

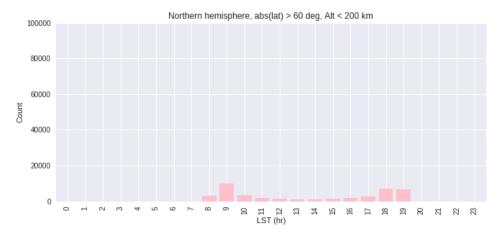
# Latitude Coverage

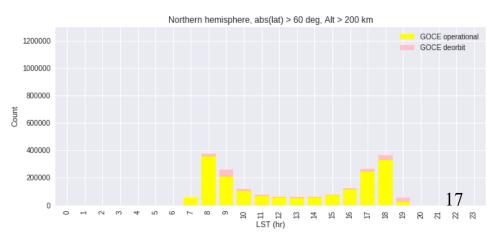


# LST coverage: Northern Hemisphere

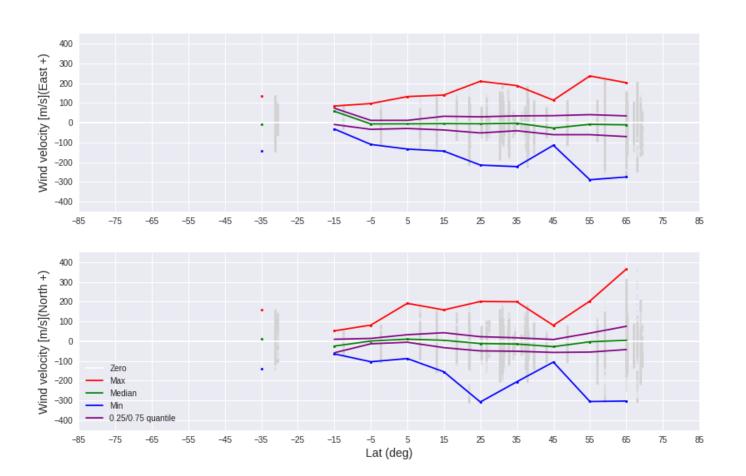




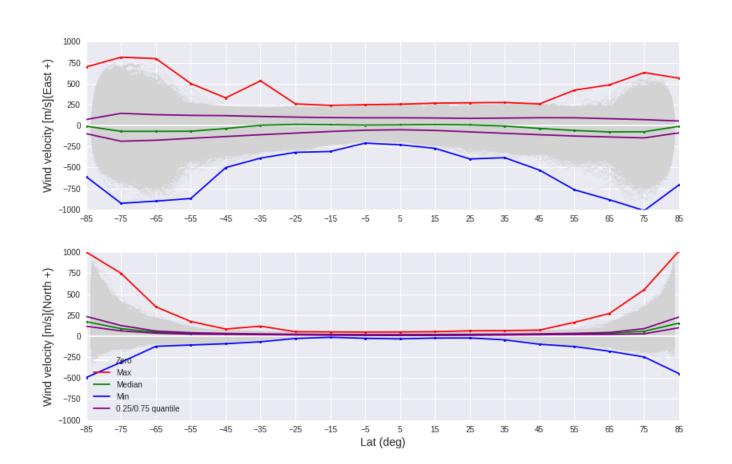




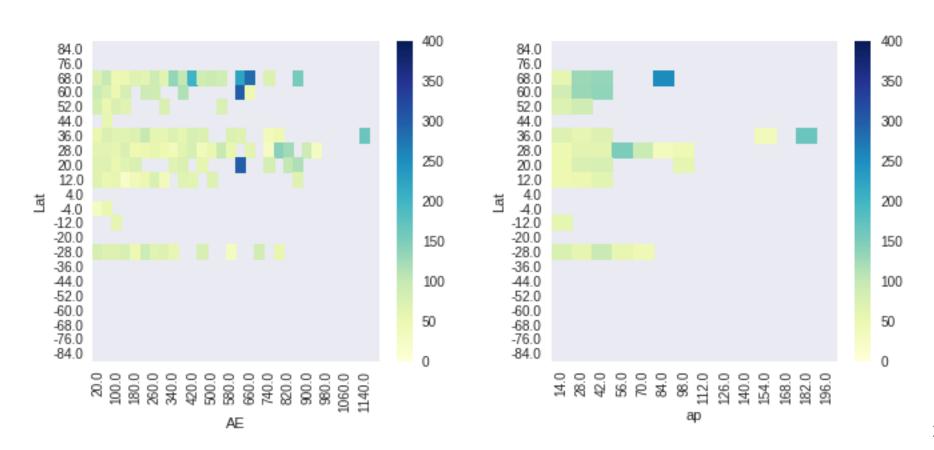
# Latitude dependency (CT)



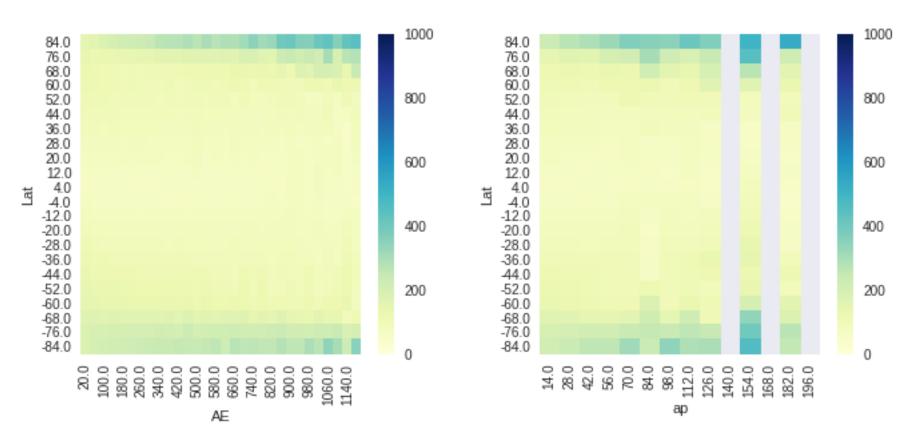
# Latitude dependency (GOCE)



### Geomag. activity dependency (CT)



# Geomag. activity dependency (GOCE)



#### Results

