

Thermosphere Density and Wind Profiles During an Auroral Substorm from the Dissipation Sounding Rocket Mission ¹Diana J Swanson, ¹James H Clemmons, ¹Lance Davis, ²Mehdi Benna, ²Robert Pfaff, ³Mark Conde, ³Don Hampton

Introduction

The comingled ionosphere/thermosphere (IT) acts as the transition region between Earth's atmosphere and space. In the high latitude IT energy flowing from the Sun and magnetosphere is dissipated through processes like Joule heating. Joule Heating is a frictional heating due to opposing motion of neutral gas and ions. This low altitude heating is one mechanism that drives neutral gas vertically, called neutral upwelling.

The Dissipation Sounding Rocket mission will answer questions regarding thermospheric wind, density and composition dynamics and structures driven by Joule heating due to electromagnetic forcing during active aurora.



Launch Conditions

Dissipation launched into a well-developed auroral display on November 8, 2023, with an apogee of 321 km. The rocket was launched slightly after the onset of a substorm allowing time for the thermosphere to heat and react to the enhanced geomagnetic activity.



Dissipation T₀ **launch time.** (Magnetometer data, Geophysical Institute, UAF 2023. Retrieved from Research Computing Systems 25 May 2023.)

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Thermosphere Density and Temperature

The Ionization Gauge (IG) on Dissipation is a Bayard Alpert IG to measure number density (pressure) of neutral gas.



The accommodation chamber allows for analysis of the number density and temperature of the ambient gas based on kinetic theory with few assumptions.

Assuming flux balance and accounting for the ram velocity :

$$n_a \sqrt{T_a} = n_g \sqrt{T_g \left\{ e^{-S^2} + S \sqrt{\pi} [1 + \operatorname{erf}(S)] \right\}}^{-}$$
$$S = \frac{|v| \cos(\alpha_{att})}{\sqrt{\frac{2k_B T_a}{\overline{m}}}}$$

Where n_a is ambient density, T_a is ambient temperature, n_a is gauge density, T_g is gauge temperature, α_{att} angle of attack of the rocket relative to the IG aperture.

A "hypsometric" approach assuming hydrostatic equilibrium of a stratified fluid according to

$$\int_{p(z_1)}^{p(z_2)} \frac{dp}{p} = \int_{z_1}^{z_2} -\frac{g}{R T} dz$$

is used to iterate the values for ambient temperature and density to calculate both density and temperature profiles.



Figure 10. Measured IG pressure (black), modulation is due to rocket spin and wind. The theoretical (red) curve represents the expected IG pressure if no wind was present.

Wind Analysis

Analysis of wind measurements from the IGs is ongoing. Due to the payload orientation of the IGs on a spinning payload *In-situ* neutral wind can be measured. IG pressure is modulated as the payload spins No Wind >> Pressure peak when IG aligned with ram Wind Pressure peak is a superposition of the rocket velocity and the wind velocity.

Constrains magnitude of wind to a line

2. Vertical winds are negligible

Figure 9. Schematic of Wind Triangulation Calculation. The resulting wind vector points to the intersect of these planes, drawn in orange





achieve the science goals of this mission.